WHAT IS CLAIMED IS:

- 1. An article comprising
- a substrate comprising silicon;
- a bond coat on the substrate, the bond coat comprising silicon;
- an intermediate coat on the bond coat, the intermediate coat comprising at least one of tantalum aluminate and niobium aluminate; and
 - a top coat on the intermediate coat.
- 2. The article of claim 1, wherein at least one of tantalum aluminate and niobium aluminate comprises about 95 to about 100 mol % of the intermediate coat.
 - 3. The article of claim 1, wherein the intermediate coat further comprises alumina.
- 4. The article of claim 3, wherein the alumina is present from about 0.1 to about 5 mol%.
- 5. The article of claim 1, wherein the intermediate coat further comprises at least one of tantalum oxide and niobium oxide.
- 6. The article of claim 5, wherein at least one of tantalum oxide and niobium oxide is present in about 0.1 to about 5 mol%.
- 7. The article of claim 1, wherein the intermediate coat consists essentially of at least one of tantalum aluminate and niobium aluminate.
- 8. The article of claim 1, wherein the intermediate coat is substantially inert to SiO_2 at temperatures greater than 1000 °C.
- 9. The article of claim 1, wherein the coefficient of thermal expansion of the intermediate coat is within about twenty percent of the coefficient of thermal expansion of the top coat.

- 10. The article of claim 1, wherein the top coat comprises barium strontium aluminosilicate.
- 11. The article of claim 1, wherein the substrate comprises at least one of silicon carbide and silicon nitride.
- 12. The article of claim 11, wherein the substrate comprises at least one of silicon carbide and silicon nitride dispersed in a matrix material.
 - 13. The article of claim 1, wherein the substrate comprises a silicon carbide matrix.
 - 14. The article of claim 1, wherein the substrate comprises a silicon carbide reinforcement in a silicon carbide matrix.
 - 15. The article of claim 1, wherein the substrate comprises a silicon carbide reinforcement in a silicon/silicon carbide matrix.
 - 16. The article of claim 1, wherein the article is a component of a gas turbine engine.
- 17. The article of claim 1, further comprising a layer of silica between the bond coat and the intermediate coat.
- 18. A method of making an article comprising applying a bond coat comprising silicon to a substrate comprising silicon; applying an intermediate coat on the bond coat, the intermediate coat comprising at least one of tantalum aluminate and niobium aluminate; and applying a top coat on the intermediate coat.
- 19. The method of claim 18, wherein the intermediate coat is applied by thermal spraying.
- 20. The method of claim 18, wherein the intermediate coat consists essentially of at least one of tantalum aluminate or niobium aluminate.

- 21. The method of claim 18, wherein the top coat comprises barium strontium aluminosilicate.
- 22. The method of claim 18, further comprising applying a layer of silica between the bond coat and the intermediate coat.
 - 23. An article comprising
 - a substrate comprising silicon;
 - a bond coat on the substrate, the bond coat comprising silicon;
 - an intermediate coat;
 - a layer comprising silica between the bond coat and the intermediate coat; and
- a top coat on the intermediate coat, wherein the intermediate coat resists a solid-state subsurface reaction between the intermediate coat and the silica.